



Outline

- Impact Testing
- Charpy Test
- Ductile to Brittle Transition Temperature
- Metallurgical Factors Affecting T_T
- Fatigue
- Fatigue Testing
- Types of Fatigue



Impact Testing

- *Charpy and Izod* tests measure *impact energy* or *notch toughness*
- Charpy V- notch (CVN) most common

Before fracture mechanics - impact testing was used to measure impact behaviour and likelihood of brittle fracture. Developed to detect the onset of brittle failure in ductile materials e.g. steel ships, bridges etc.

- Still used in quality control and Standards (ship plate etc).

Three main factors were producing these fractures in service:

- Triaxial stress state (at notches, cracks etc)
- Low temperatures
- High strain or loading rates

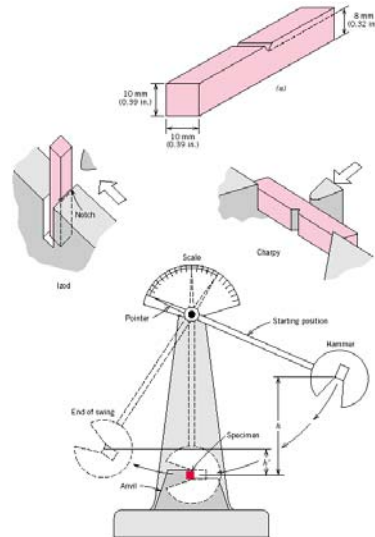
Impact testing is used for:

- ✓ checking quality
- ✓ tendency for brittle failure
- ✓ temperature dependence.



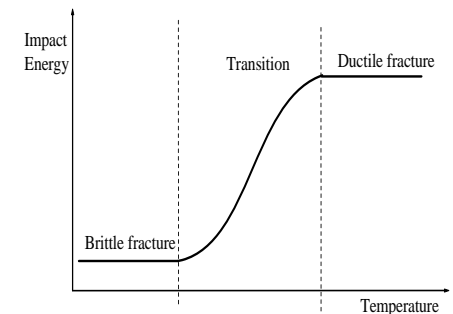
Charpy Test

- Use *standard sized bar* specimens with a central notch
- Weighted pendulum released from a height, h
- Impacts the specimen behind the notch (stress concentration)
- Fracture of specimen occurs and energy is absorbed
- The pendulum travels to point, h' , where $h' < h$
- Obtain the amount of absorbed energy from scale
- and test method



Ductile to Brittle Transition in Steel

- Primary function of Charpy test
- At high temperature, CVN for steel is relatively high but drops with decrease in temperature
- At **low temperature** steel can be **brittle**
- The sudden drop in impact energy is the **transition (DBT)**
- Steels should always be used **above** their DBT
- Polymers also experience DBT
- Aluminum and copper alloys show



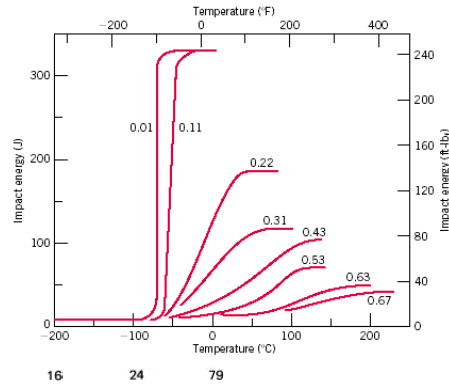
Al and Cu have FCC structure



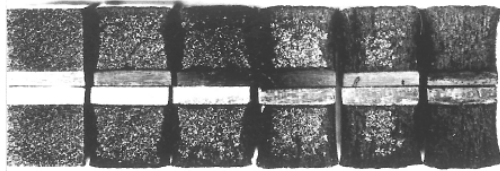
Ductile to Brittle Transition Behavior

- Actual DBT is to define, instead minimum requirement of CVN = 20J (15ft.lb) is used
- DBT changes with carbon content in steel
- Steels: 0.01-0.67% carbon

Can use fracture surface appearance to estimate the DBT



Brittle, shiny, faceted, bright, flat overall, no or little deformation



Fibrous, grey, dull, possibly ridged. Sides may be pulled in. Hinged.

Figure 6-11 (Hertzberg)

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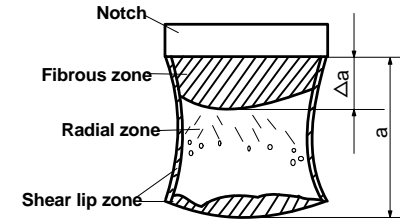
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Ductile to Brittle Transition Behavior

- Like tensile specimen, the fractured surface of an impact specimen also contains fibrous, radial (*granular and shiny zone*) and shear lip zones.
- Fibrous zones occur during the process of crack propagation in materials.
- The proportions of fibrous zone, radial zone and shear lip zone vary with temp.



A sketch of the appearance of fractured surface of impact specimen after Charpy test

- When temp. drops to a certain value, the area of the fibrous zone and the area of granular zone sharply. → The fracture behavior of the material transforms from ductile to brittle.
- DBT temp is usually determined as the area of crystallization zone accounts for 50% of the whole fractured surface (*i.e. as the temp. at which the failure surface is 50% shiny*), and it is denoted by $FATT_{50}$ (Fracture Appearance Transition Temp.)

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Selecting Transition Temperature

- T_1 : Conservative, above T_1 fracture is 100% fibrous. *Fracture Transition Plastic* (FTP) - very demanding.
- T_2 : 50% cleavage - 50% ductile *Fracture Appearance Trans. Temp.* (FATT).
- T_3 : Average of upper and lower shelf values. (often approx = T_2)
- T_4 : Arbitrary value of energy absorbed, (CVN) e.g. 20 J (15 ft.lb) for low strength ship steel. *Ductility Transition Temp.*
- T_5 : 100% cleavage fracture. *Nil Ductility Temperature* (NDT)

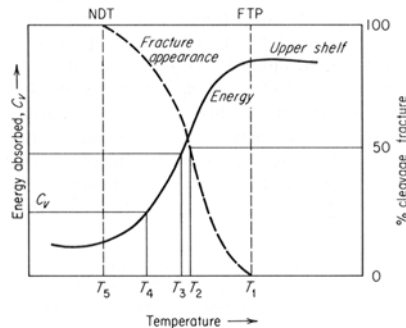


Figure 14-6. Various criteria of transition-temperature obtained from Charpy tests

The lower this temperature, the greater the fracture toughness

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Charpy Test

An instrumented Charpy test allows determination of energy required to crack and also energy to crack rather than just total energy for fracture.

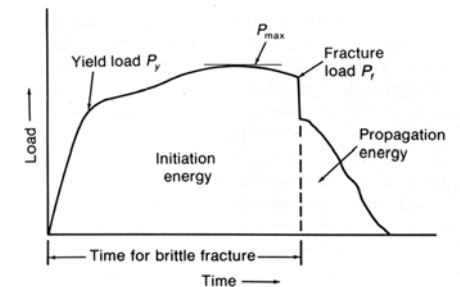
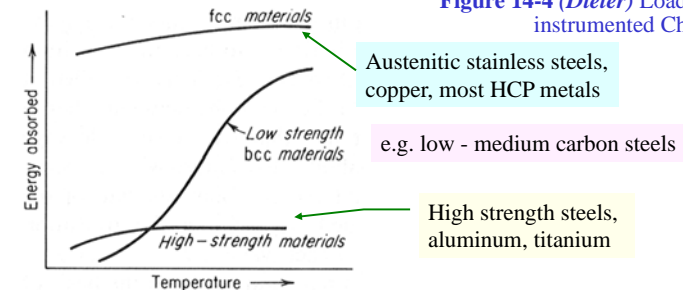


Figure 14-4 (Dieter) Load-time history for an instrumented Charpy test.



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Metallurgical Factors Affecting T_T

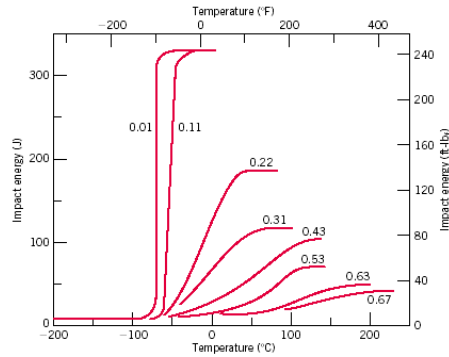
➤ For steels: As %C ↑ ⇒ σ_y ↑, σ_{TS} ↑, H ↑, % El ↓, CVN ↓ and T_T ↑

- This can be countered by adding Manganese - Mn : C should be 3:1

➤ Phosphorous increases T_T , Oxygen in steel T_T :

- semi-killed (add Si) and

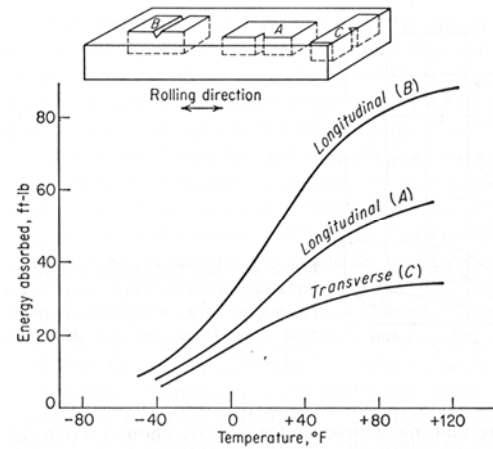
- fully-killed (add Si + Al) to remove oxygen



Remember also: as grain size ↓ toughness and T_T ↓.
Niobium and vanadium added to keep grain size small.



Metallurgical Factors Affecting T_T



➤ Rolled and forged products may have varying impact behaviour due to grain orientation.

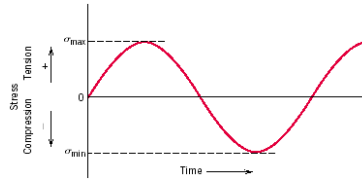
➤ Note that the difference is not as large at lower temperatures.

Figure 14-8 (Deiter) Effect of specimen orientation on Charpy transition-temperature curves.



Fatigue

- Occurs under *dynamic or fluctuating* stresses
- Examples:** bridges, automobiles, aircraft and machine components
- Failure can occur at stress than under *static loading*
- Accumulated damage (cracking) occurs over a long period of time → *catastrophic failure*
- 90%** of metal failures due to mechanical causes occur in **fatigue!**
- Also occurs in *ceramics, polymers* and composites
- Appears as **brittle-like** failure even in ductile materials
- Usually breaks **without warning**; no or very little, observable plastic deformation (*some micro-deformation*).



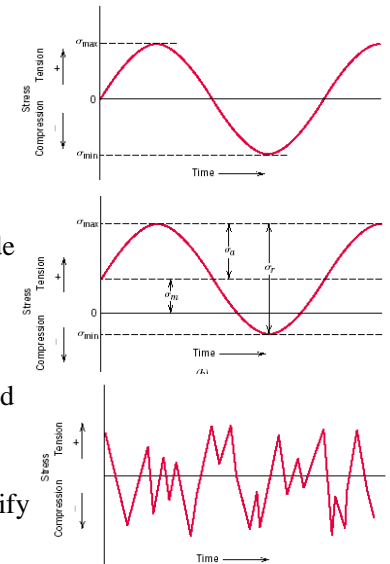
Fatigue consists of two stages:

- Crack Initiation
- Crack Propagation



Cyclic Stresses

- Applied stress:
 - axial (compression-tension)
 - flexural (bending)
 - torsional (twisting)
- Reversed stress cycle** is where the sinusoidal stress is of equal amplitude
 - ➔ about a mean of zero
 - σ_{max} is tensile and σ_{min} is compressive
- Repeated stress** occurs when σ_{max} and σ_{min} are about $\sigma = 0$
- Random stress** often occurs in engineering and is less easy to quantify





Cyclic Stresses

- Mean stress, σ_m is defined as:

$$\sigma_m = \frac{\sigma_{\max} + \sigma_{\min}}{2}$$

- *Range of stresses* (σ_r) is the difference between σ_{\max} and σ_{\min} :

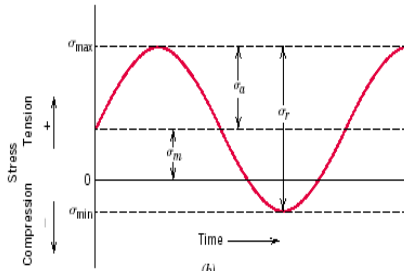
$$\sigma_r = \sigma_{\max} - \sigma_{\min}$$

- *Stress amplitude*, σ_a :

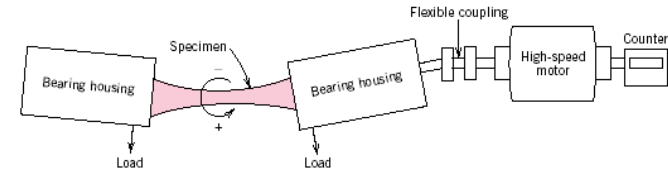
$$\sigma_a = \frac{\sigma_r}{2} = \frac{\sigma_{\max} - \sigma_{\min}}{2}$$

- Ratio of max. and min. stress is the *stress ratio*, R:

$$R = \frac{\sigma_{\min}}{\sigma_{\max}}$$



Fatigue Testing and S-N Curves



- Laboratory simulation using *rotating beam test*.
- It creates a reverse cycle bending with rotation (compression/tension)
- Apply a stress of $\sigma_{\max} \sim 2/3 \sigma_{TS}$
- Measure number of cycles (N) to failure
- Repeat using progressively *lower* σ_{\max} (S)
- Plot S vs. log N → S-N curve



Fatigue Testing and S-N Curves

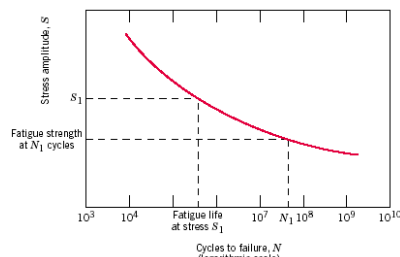
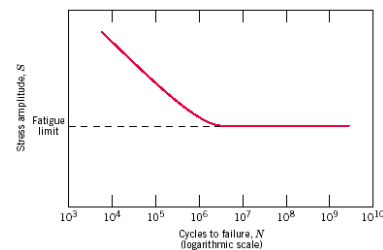
Two types of S-N curve

1. *Fatigue (endurance) limit* for ferrous and titanium alloys
- FL = 0.35 - 0.6 σ_{TS} (typically)

2. *No fatigue limit* with alloys (Al, Mg and Cu)

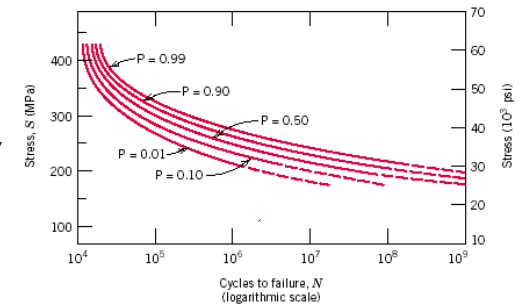
→ Need to define a fatigue strength at a specified number of cycles (10^7)

→ *Fatigue life* (N_f) is the number of cycles at a specified stress level (S_f)



Statistical Nature of Fatigue

- The data on S-N curves are scattered due to *material variability* and test parameters are *difficult to control*
- Fatigue strengths are usually average values
- Probability of failure (P) defined, e.g. at 215MPa 1% of samples fail at 10^6 cycles



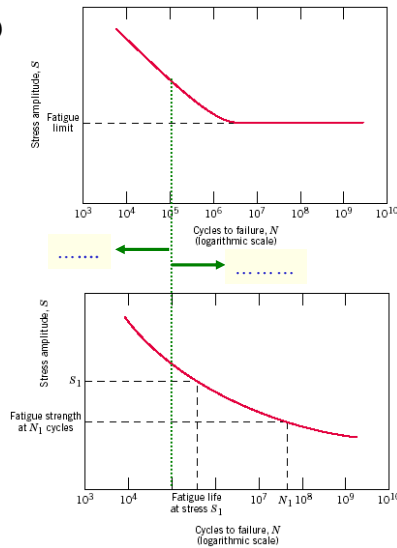
Since, this type of curves shows of fatigue failure at certain stress level, it is more accurate than “average” value as is normally shown.



Types of Fatigue

Low cycle fatigue

- high loads → N_f (10^4 - 10^5 cycles)
- high stress environment with high design stress and small safety factor, $DS \approx \sigma_{ys}$
- scheduled inspection and maintenance of parts (aircraft)
- most common cause of fatigue cracking and failure



High cycle fatigue

- low loads → N_f ($>10^5$)
- typically involves low design stresses, $DS \ll \sigma_{ys}$
- **less common** cause of failure, results from poor design or environmental effects



Low-Cycle Fatigue

- This type of cycling is more likely in **nuclear pressure-vessels**, **steam turbines**, and similar components where repeated stresses are created by thermal fluctuations.
- e.g. if material is constrained and then heated, **thermal stresses** are generated; if it is not constrained we have cyclic **thermal strains** (i.e. heating and cooling of vessels).

- *Fatigue results from cyclic*: this is induced by the restraint of the dimensional expansion and/or contraction occurring due to varying the temperature.



Low-Cycle Fatigue

Coffin-Manson (1950's) relation: $\frac{\Delta \epsilon_p}{2} = \epsilon'_f (2N)^C$

where $\Delta \epsilon_p/2$ = plastic strain amplitude

ϵ'_f = fatigue ductility coefficient \approx true strain at fracture, ϵ_f

C = fatigue ductility exponent

(-0.5 to -0.7 for most metals; lower C value = longer life)

- Stress is high enough for plastic deformation to take place
- Strain is more representative than stress in this case

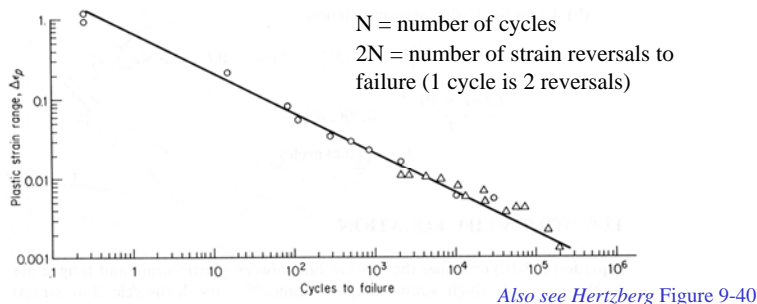


Figure 12-13 Low-cycle fatigue curve ($\Delta \epsilon_p$ vs. N) for Type 347 stainless steel. (From L. F. Coffin,

Also see Hertzberg Figure 9-40



Next time:
Continue Fatigue